



Transportation
Safety Board
of Canada

Bureau de la sécurité
des transports
du Canada



AIR TRANSPORTATION SAFETY INVESTIGATION REPORT A23W0048

HARD LANDING

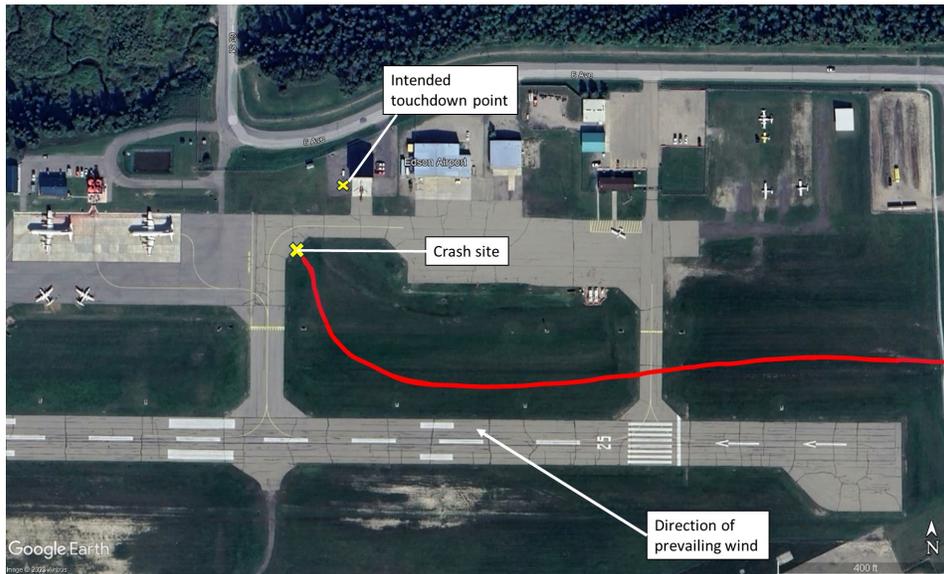
Range Helicopters Inc.
Airbus Helicopters AS350 B2 (helicopter), C-FAVX
Edson Airport (CYET), Alberta
04 May 2023

The Transportation Safety Board of Canada (TSB) investigated this occurrence for the purpose of advancing transportation safety. It is not the function of the Board to assign fault or determine civil or criminal liability. **This report is not created for use in the context of legal, disciplinary or other proceedings.** See the Terms of use at the end of the report.

History of the flight

On 04 May 2023, the Range Helicopters Inc. (Range Helicopters) Airbus Helicopters AS350 B2 helicopter (registration C-FAVX, serial number 4242) was conducting a return flight to Edson Airport (CYET), Alberta, from a forest firefighting staging area located approximately 49 nautical miles (NM) to the east. When the helicopter was 5 NM east of CYET, a flight of 4 Canadair CL-215T aircraft was approaching the airport from the west and intending to land on Runway 07, which was active at the time. It was reported that the wind was strong and gusting from the east-southeast. The occurrence pilot's usual approach to the Range Helicopters hangar at CYET involved passing to the south of the town of Edson, Alberta, and then approaching the airport from the west. Upon hearing the intentions of the flight of CL-215T aircraft, the pilot, not wanting to delay the group's arrival, decided to approach the airport downwind and to the north of Runway 07, between the runway and the taxiway, with the intent of conducting a tight right turn to align the helicopter into the wind (Figure 1).

Figure 1. Occurrence helicopter's flight path to Edson Airport (Source: Google Earth, with TSB annotations)



As the helicopter approached between Runway 07 and the grassy area between taxiways A and B, the pilot began to slow down, eventually reaching a groundspeed of 30 knots.¹ He noticed a pair of air tankers at the fire base ahead and to his right, and, not wanting to overfly them, he slowed down even more. The helicopter was in a descent at a rate between 300 and 500 fpm with a tailwind of 27 to 36 knots. Just before entering the right turn toward the hangar, the helicopter began an uncontrolled descent from between 50 and 30 feet above ground level (AGL). The pilot applied collective to reduce the rate of descent; however, the rate of descent increased. This scenario is consistent with a vortex ring state (VRS).² The aircraft struck the ground and rolled onto its right side at 2005.³ The pilot sustained minor injuries. He was able to extricate himself from the wreckage. There was no fire. The emergency locator transmitter⁴ activated.

Pilot information

The pilot held a commercial pilot licence – helicopter and a valid Category 1 medical certificate. His total flight time was 1901 hours. A review of his personal log showed time as pilot-in-command in a variety of light helicopters, mainly involving pipeline patrols. He had achieved 121 hours flight time in the AS350 B2, including 118 hours as pilot-in-command.

¹ The TSB laboratory extracted flight path data from the Garmin Aera660 Global Positioning System that was on board the helicopter.

² According to the Federal Aviation Administration (FAA) of the United States' *Helicopter Flying Handbook*, "[v]ortex ring state [...] describes an aerodynamic condition in which a helicopter may be in a vertical descent with 20 percent up to maximum power applied, and little or no climb performance." (Source: Federal Aviation Administration, *Helicopter Flying Handbook* [FAA-H-8083-21B], [2019], p. 11-9.)

³ All times are Mountain Daylight Time (Coordinated Universal Time minus 6 hours).

⁴ ACR/Artex ME406HM P/N453-6604 S/N 242-06280.

In support of firefighting activity, the pilot had been on flight duty for 7 days before the occurrence during which time he flew 46.6 flight hours.⁵ During that period, he had worked 85.5 duty hours, including several 14-hour duty days.⁶ However, the company had mitigated the risk posed by the long hours by housing the crews in hotels located near the fires, allowing more time for uninterrupted rest. As part of this investigation, the TSB conducted a fatigue assessment and it was determined that fatigue was likely not a factor in this occurrence.

Company training

The pilot underwent annual recurrent training and completed all computer-based ground training and flight training by the end of March 2023. The flight training included exercises in the recognition of and recovery from VRS. These exercises were carried out at safe altitudes that allowed adequate margins for safe recoveries.

Aircraft information

Table 1. Aircraft information

Manufacturer	Eurocopter*
Type, model and registration	AS350 B2, C-FAVX
Year of manufacture	2007
Serial number	4242
Certificate of airworthiness	22 May 2007
Total airframe time	5334.2 (flight time on the day of the occurrence estimated at 7.5 hours)
Engine type (number of engines)	Turbomeca Arriel 1D1 Free turbine, turboshaft (1)
Rotor type (number of blades)	Semi-rigid (3)
Maximum allowable take-off weight	2250 kg (4961 lbs)
Recommended fuel types	Jet A, Jet A-1, Jet B
Fuel type used	Jet A

* In 2014, Eurocopter became Airbus Helicopters, which currently holds the type certificate for the aircraft type.

The Airbus Helicopters AS350 B2 is a 6-seat, single-pilot, single-turbine-powered helicopter. It has a 3-bladed counterclockwise-rotating main rotor (when viewed from below). The pilot normally flies from the right seat. The occurrence helicopter was equipped with an external cargo basket installed on the left side of the helicopter.

Based on available information about equipment and personnel on board the helicopter, the helicopter's weight and centre of gravity were within the prescribed limits.

⁵ Section 702.92 of the *Canadian Aviation Regulations* allows for a maximum of 60 hours of flight time in any 7 consecutive days.

⁶ Section 702.93 of the *Canadian Aviation Regulations* allows for a maximum of 14 consecutive flight duty hours in any 24-hour period.

The helicopter was certified and equipped for daytime visual flight rules (VFR) flight in accordance with section 605.14 of the *Canadian Aviation Regulations (CARs)*.⁷

At the time of the occurrence, there were no documented defects in the helicopter's journey log, which was recovered from the wreckage. The last entry in the journey log, dated 04 May 2023, was a daily inspection carried out by the occurrence pilot that included repetitive inspections required by an ongoing airworthiness directive.

Impact and wreckage information

The helicopter struck the ground between the runway and the taxiway in a tail-low attitude, causing the tail rotor assembly to fracture. The helicopter then rotated to the right and struck the ground forcefully enough to break the right skid. It then rolled onto its right side, and the main rotor came into contact with the ground, dragging the fuselage around in an approximate 100° arc in a counterclockwise direction. The right-forward cabin side pillar fractured, but the cabin maintained its overall shape. The engine and transmission broke free of some of their attachment points but stayed mainly in the powerplant bay. A fuel line fractured, and there was a fuel spill, but no fire (Figure 2).

Figure 2. Accident site looking north (Source: TSB)



Weather information

The aerodrome routine meteorological report issued for the Edson Airport at 2000 was as follows:

- Winds from 150° true at 27 knots, gusting to 36 knots
- Visibility of 1½ statute miles in haze
- A layer of scattered cloud based at 3000 feet AGL, with a broken ceiling at 10 000 feet AGL
- Temperature 24 °C, dew point 2 °C

⁷ Transport Canada, SOR/96-433, *Canadian Aviation Regulations*, section 605.14: Power-driven Aircraft — Day VFR.

- Altimeter setting 29.73 inches of mercury (inHg)
- Density altitude 5000 feet

There is no aerodrome forecast issued for CYET.

It was reported that although visibility during the return flight was good, it began to deteriorate approximately 10 NM east of CYET. Once the occurrence aircraft was in the area of the airport, the visibility had reduced to 1½ statute miles. The pilot was aware of the wind condition as it was being discussed by the arriving CL-215T aircraft on the aerodrome traffic frequency.

Vortex ring state

Helicopters are susceptible to an aerodynamic phenomenon known as VRS. VRS occurs when a helicopter's flight path, airspeed, and rate of descent coincide with the helicopter's downwash.⁸ In normal flight, airflow from the main rotors is directed downward. In VRS, the tip vortices generated by the main rotors re-circulate through the rotor, adversely affecting lift. Applying more power (increasing collective pitch) serves to further accelerate the downwash through the main rotor, exacerbating the condition. In a fully developed VRS, the helicopter may experience uncommanded pitch and roll oscillations, and the rate of descent may approach 6000 fpm.⁹

The *Helicopter Flying Handbook*¹⁰ published by the Federal Aviation Administration (FAA) of the United States identifies the following combination of conditions as being likely to cause VRS:

- a vertical descent or nearly vertical descent of at least 300 fpm;
- powered flight, typically within the range of 20-100% engine torque; and
- horizontal velocity slower than effective translational lift.

Airbus Helicopters advises¹¹ that to avoid VRS during approach and landings, it recommends:

When the airspeed is below 30 knots be aware of your RoD [rate of descent]. Use collective pitch (power) to control vertical speed. **NEVER ALLOW AN RoD GREATER THAN 500 fpm WHEN BELOW 30 KIAS. AVOID AGGRESSIVE DECELERATIONS WHILE IN DESCENT OR WHEN TURNING TO A DOWNWIND POSITION.** [emphasis in original]

To reduce the risk of entering VRS, helicopter pilots are trained to avoid entering their helicopter's own downwash. Should pilots of single main-rotor helicopters find themselves in VRS,

Transport Canada recommends the following 2 recovery methods:

- **Dive Out** – Apply forward cyclic while reducing the collective in an attempt to gain airspeed. As airspeed increases, the helicopter will move out of its downwash and normal flight can be resumed.

⁸ Transport Canada, *Helicopter Flight Training Manual* (TP 9982), Second Edition (June 2006), p. 107.

⁹ Federal Aviation Administration, *Helicopter Flying Handbook* (FAA-H-8083-21B) (2019), p. 11-9.

¹⁰ *Ibid.*, p. 11-10.

¹¹ Airbus Helicopters, Safety Information Notice No. 3123-S-00; Useful information about the Vortex Ring State (VRS) phenomenon, Revision 1 (12 April 2022), p. 11.

- **Enter Autorotation** – This results in changing the airflow from disturbed airflow due to VRS to upward autorotational airflow. The pilot may then ease the cyclic forward, gain airspeed, and increase power to resume normal flight.¹²

In recent years, a new recovery technique, the Vuichard Recovery, has gained popularity. This technique involves “lateral cyclic thrust combined with an increase in power and lateral antitorque thrust [...] eliminating the descent rate as opposed to exiting the vortex.”¹³

Typically, a helicopter in VRS will lose considerable altitude before it is able to resume normal flight. If insufficient altitude is available, the helicopter may impact the ground before it is able to recover from VRS.

Ideally, helicopters should be positioned into the wind for final approach to reduce the risk of entering VRS. A pilot conducting a downwind approach must carefully manage the helicopter’s ground speed, power, and rate of descent to prevent the helicopter from descending into its own downwash and being caught in it, possibly at an altitude from which recovery is impossible.

TSB laboratory reports

The TSB completed the following laboratory report in support of this investigation:

- LP074/2023 – NVM Recovery – GPS and Flight Tracker

Safety messages

This investigation highlights the importance of approach planning and power management, particularly when conducting a downwind approach to a landing site. Therefore, it is critical that helicopter pilots remain aware of and avoid flight conditions that can cause VRS to develop. Any potential time saved by attempting a downwind approach must be carefully weighed against the risks that it may present.

This report concludes the Transportation Safety Board of Canada’s investigation into this occurrence. The Board authorized the release of this report on 21 November 2023. It was officially released on 29 November 2023.

Visit the Transportation Safety Board of Canada’s website (www.tsb.gc.ca) for information about the TSB and its products and services. You will also find the Watchlist, which identifies the key safety issues that need to be addressed to make Canada’s transportation system even safer. In each case, the TSB has found that actions taken to date are inadequate, and that industry and regulators need to take additional concrete measures to eliminate the risks.

¹² Transport Canada, *Helicopter Flight Training Manual* (TP 9982), Second Edition (June 2006), p. 107.

¹³ Federal Aviation Administration, *Helicopter Flying Handbook* (FAA-H-8083-21B) (2019), p. 11-10.

ABOUT THIS INVESTIGATION REPORT

This report is the result of an investigation into a class 4 occurrence. For more information, see the Policy on Occurrence Classification at www.tsb.gc.ca

The Transportation Safety Board of Canada (TSB) investigated this occurrence for the purpose of advancing transportation safety. It is not the function of the Board to assign fault or determine civil or criminal liability.

TERMS OF USE

Use in legal, disciplinary or other proceedings

The *Canadian Transportation Accident Investigation and Safety Board Act* states the following:

- 7(3) No finding of the Board shall be construed as assigning fault or determining civil or criminal liability.
- 7(4) The findings of the Board are not binding on the parties to any legal, disciplinary or other proceedings.

Therefore, the TSB's investigations and the resulting reports are not created for use in the context of legal, disciplinary or other proceedings.

Notify the TSB in writing if this investigation report is being used or might be used in such proceedings.

Non-commercial reproduction

Unless otherwise specified, you may reproduce this investigation report in whole or in part for non-commercial purposes, and in any format, without charge or further permission, provided you do the following:

- Exercise due diligence in ensuring the accuracy of the materials reproduced.
- Indicate the complete title of the materials reproduced and name the Transportation Safety Board of Canada as the author.
- Indicate that the reproduction is a copy of the version available at [URL where original document is available].

Commercial reproduction

Unless otherwise specified, you may not reproduce this investigation report, in whole or in part, for the purposes of commercial redistribution without prior written permission from the TSB.

Materials under the copyright of another party

Some of the content in this investigation report (notably images on which a source other than the TSB is named) is subject to the copyright of another party and is protected under the *Copyright Act* and international agreements. For information concerning copyright ownership and restrictions, please contact the TSB.

Citation

Transportation Safety Board of Canada, *Air Transportation Safety Investigation Report A23W0048* (released 29 November 2023).

Transportation Safety Board of Canada
200 Promenade du Portage, 4th floor
Gatineau QC K1A 1K8
819-994-3741; 1-800-387-3557
www.tsb.gc.ca
communications@tsb.gc.ca

© His Majesty the King in Right of Canada, as represented by the Transportation Safety Board of Canada, 2023

Air transportation safety investigation report A23W0048

Cat. No. TU3-10/23-0048E-PDF

ISBN: 978-0-660-68642-4

This report is available on the website of the Transportation Safety Board of Canada at www.tsb.gc.ca

Le présent rapport est également disponible en français.